iss_rad, radgse at LAB1O2, ER1, Lockers 3,4:[135.28 -10.68 132.12] 0.0625 sa/sec (1.00 Hz)

SSAnalysis[0.0 0.0 0.0]



2014 Dragon-3 Capture and Install Qualify

	Description							
Senso	ISS radgse 0.0625 sa/sec, 1.0 Hz							
Location	ISS							
Plot Type	e Acceleration vs. Time							
 Plot Type Acceleration vs. Time Notes (all times are GMT hh:mm): This 3-panel plot of XYZ acceleration versus time shows the primary impact to the microgravity environment associated with the capture and install of Dragon-3 cargo vehicle. From 08:56 to 09:01, the ISS did a maneuver to capture attitude. This is seen as a sudden step down on the Z-axis. Next, at 11:05 they went to attitude hold with desats (thrusters) inhibited, with SpaceX Dragon-3 free flying to Node 2 nadir port. At 11:14, the ISS robotic arm was used to capture the Dragon-3 cargo ship. From 11:40 to 11:45, the ISS did a maneuver to Dragon-berthed attitude. At 12:46, the ISS went to momentum management for attitude control. From 13:00 to 14:26, ISS thrusters were disabled to allow for Dragon-3 install via robotic arm 								
disabled to allow for Dragon-3 install via robotic arm								
Regime:	Vibratory							
Category: Vehicle								



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2014 Dragon-3 Capture and Install

Source:

iss_rad, radgse at LAB1O2, ER1, Lockers 3,4:[135.28 -10.68 132.12] 0.0625 sa/sec (1.00 Hz)

SSAnalysis[0.0 0.0 0.0]



2014 Dragon-3 Capture and Install Quantify

Description								
Sensor ISS radgse 0.0625 sa/sec, 1.0 Hz								
Location	ISS							
Plot Type	Acceleration vs. Time							
 Note type Acceleration vs. time Notes: This 3-panel plot of XYZ acceleration versus time is similar to the previous page, except we zoom out in time in order to better show the quantitative difference in the quasi-steady microgravity environment. Note before GMT 20-Apr-2014, 06:00 that the Z-axis component of the quasi-steady acceleration vector was about -0.18 ug. After the Dragon-3 was installed, we see a step up of the Z-axis component of the quasi-steady acceleration vector to about -0.105 ug. The difference, therefore, was a Z-axis shift of about 0.075 ug. If you look closely, you will notice that there is also a minor shift on the X-axis too when we compare before versus after Dragon-3 install. 								

	Regime:	Vibratory
	Category:	Vehicle
	Source:	2014 Dragon-3 Capture and Install





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2014 Dragon-3 Capture and Install Ancillary Notes

The table below shows the as-flown timeline of events leading up to Dragon-3 capture on GMT 20-Apr-2014 at 11:14 via robotic arm. This was followed by install between 13:00 and 14:26. As seen in the plots on the previous pages, the main impact on the microgravity environment was the maneuver to capture attitude, the maneuver to Dragon-berthed attitude, and the resultant shift of the quasi-steady (steady state) Z-axis acceleration vector component of about 0.075 ug.

Dragon-3 Capture (M14_110_A_03.UAF)	GMT	Att. Name	YPR	F/T Cfg	Event	4/20/2014
15	110/08:56	+XVV	356	MMT	Transition to USTO	
	—	+ZLV	357.1	UST		
		TEA	0.6			
16	110/08:56	+XVV	356	UST	Maneuver to Capture Attitude	
	110/09:01	+ZLV	356	UST		
			1			
17	110/11:05	+XVV	356	UST	Transition to Att Hold with Desats Inhibit (SpaceX FF to N2 Nadir)	Capture at 11:14
	_	+ZLV	356	AHC		
10	110/11.20	1 1/1 // /	250	ALIC	Transition to LICTO	
18	110/11:20	+XVV	350	AHC	Transition to USTO	
		⊤ZLV	330	051		
10	110/11.40	+ VW	256	UST	Manauwar to Dragon both of TEA w/ DEADL270 SEADL00	
15	110/11.40	+71 V	358.1	UST	Maneuver to Diagon bernieu TEA w/ PSAKJ 270, SSAKJ 50	
	110/11.45	TEA	0.6	051		
20	110/12.46	+XVV	356	UST	Transition to Momentum Management using USTO	TEA for VV#3z N2nDze_PSAR1270_SSAR190
		+ZLV	358.1	SAT		, , , , , , , , , , , , , , , , ,
		TEA	0.6			
Dragon-3 Install (M14_110_B_03.UAF)						4/20/2014
21	110/13:00	+XVV	356	SAT	Disable Thrusters	
	110/14:26	+ZLV	358.1	SAT		
		TEA	0.6			
22	110/14:26	+XVV	356	SAT	Enable Thrusters	
	_	+ZLV	358.1	MMT		
		TEA	0.6			







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